



# **Redesign of the VAPoR Miniaturized Pyrolysis TOFMS for Improved Sensitivity**

**Adrian Southard**

NASA Goddard Space Flight Center and University  
Space Research Agency

Additional authors: **Stephanie A. Getty, Carl Kotecki,  
Steve Feng, Danny Glavin (PI)**

HEMS Workshop September 19-22, 2011

# Goddard Space Flight Center

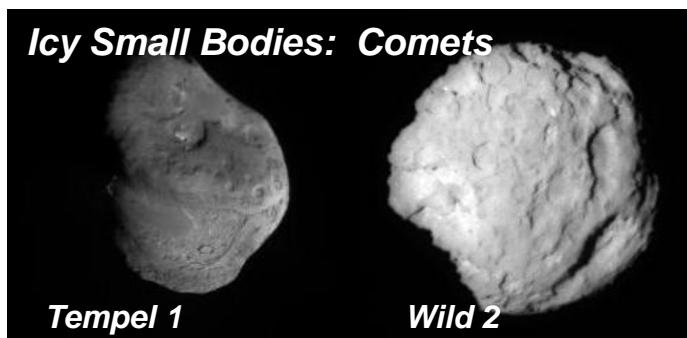
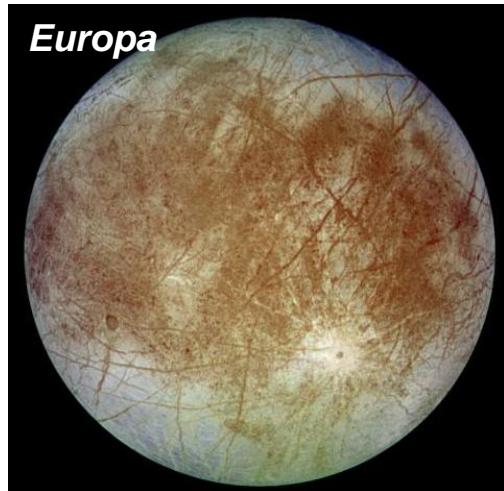
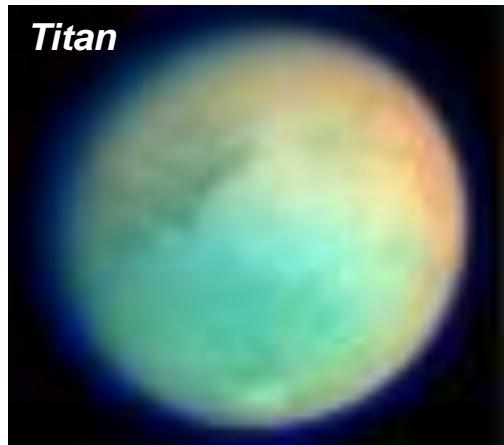


NASA Centers (10):

- + Am
- + Dry
- + Gle
- + Go
- + Jet
- + Joh
- + Kee
- + Lar
- + Ma
- + Ste



# Planetary Targets for *In Situ* Instruments



**Saturn**

**Icy Small Bodies: Asteroids**

**Eros**

**Icy Small Bodies: Comets**

**Tempel 1**

**Wild 2**

# Planetary Mass Spec: State of the Art

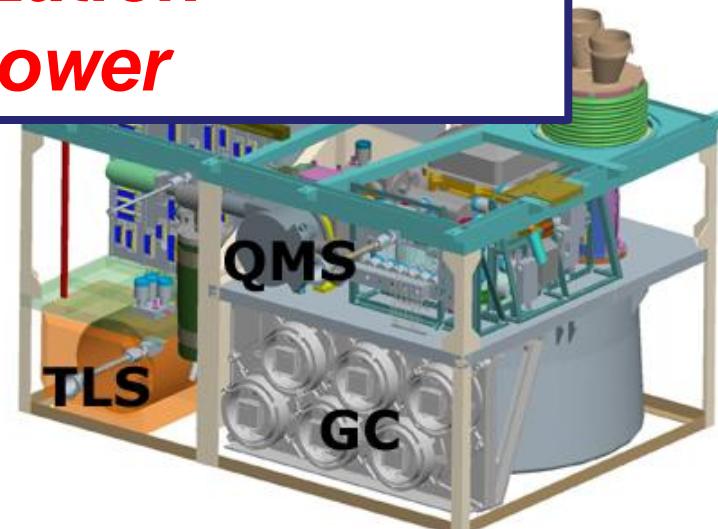


Pyrolysis Gas Chromatograph  
Quadrupole Mass Spectrometer  
– part of Sample Analysis at Mars  
Instrument Suite

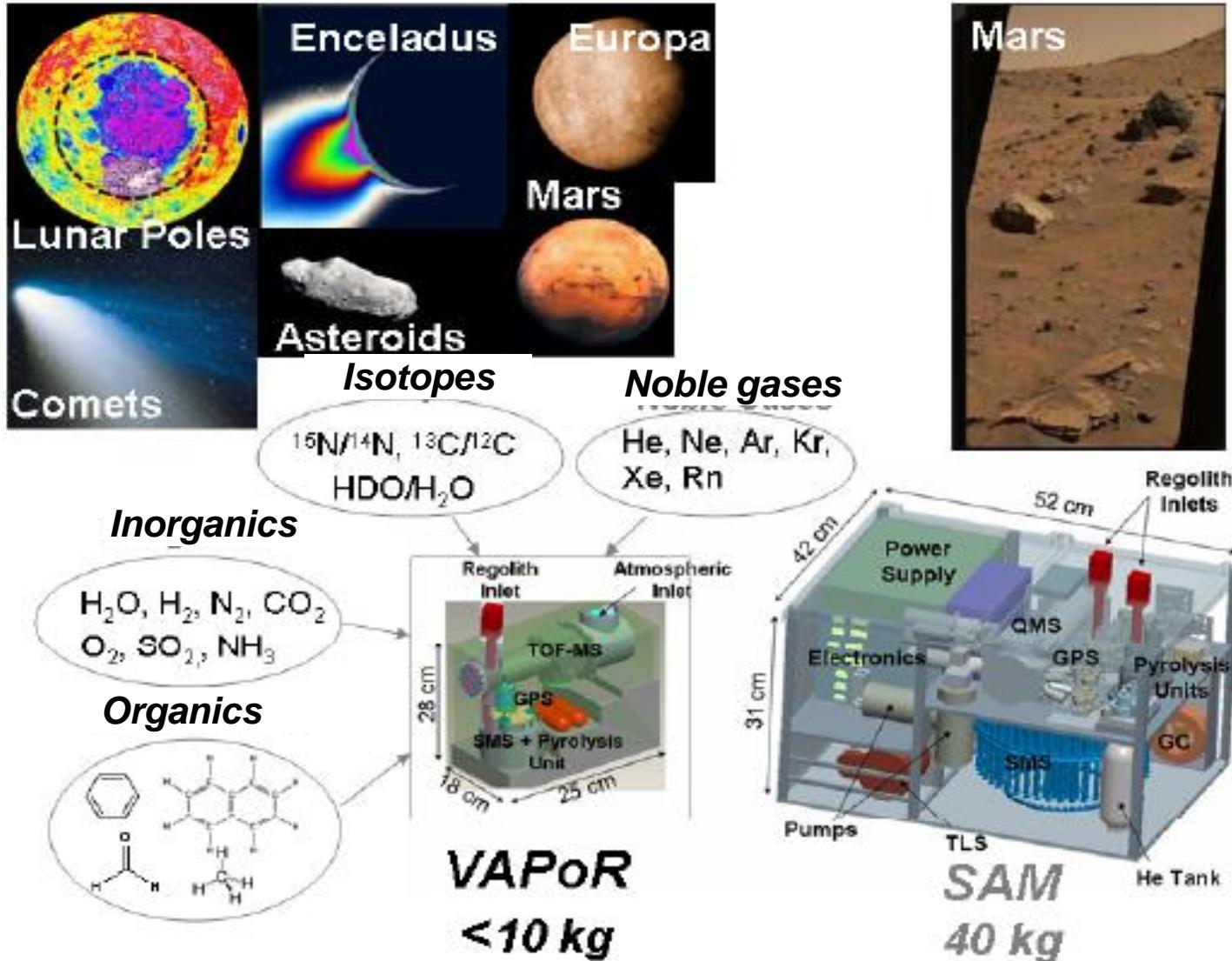
## *Opportunities for improvement:*

- 1) Miniaturization**
- 2) Lower power**

- QMS similar to Huygens
  - m ~ 1.3 kg, P ~14.5 W
- Thermionic filaments are used to ionize pyrolysis products or atmospheric gases



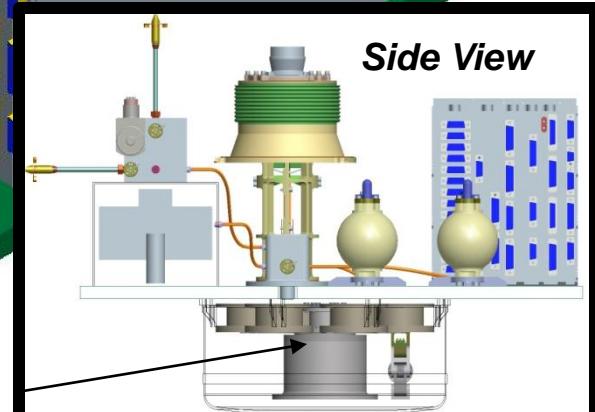
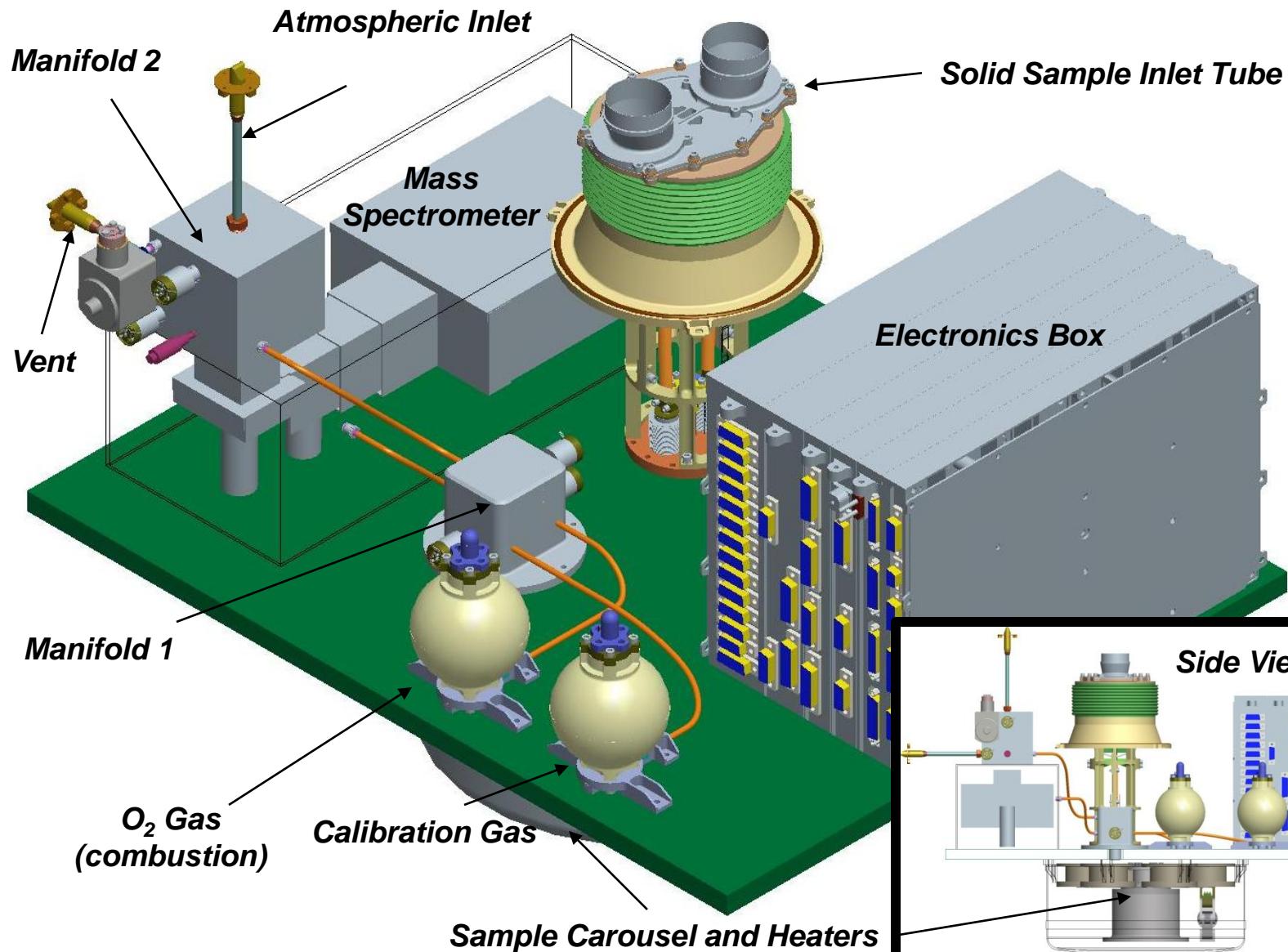
## VAPoR – Volatile Analysis by Pyrolysis of Regolith



# VAPoR Key Components



V. Holmes (Mechanical Design)



# VAPoR Measurements of Lunar Atmosphere and Regolith

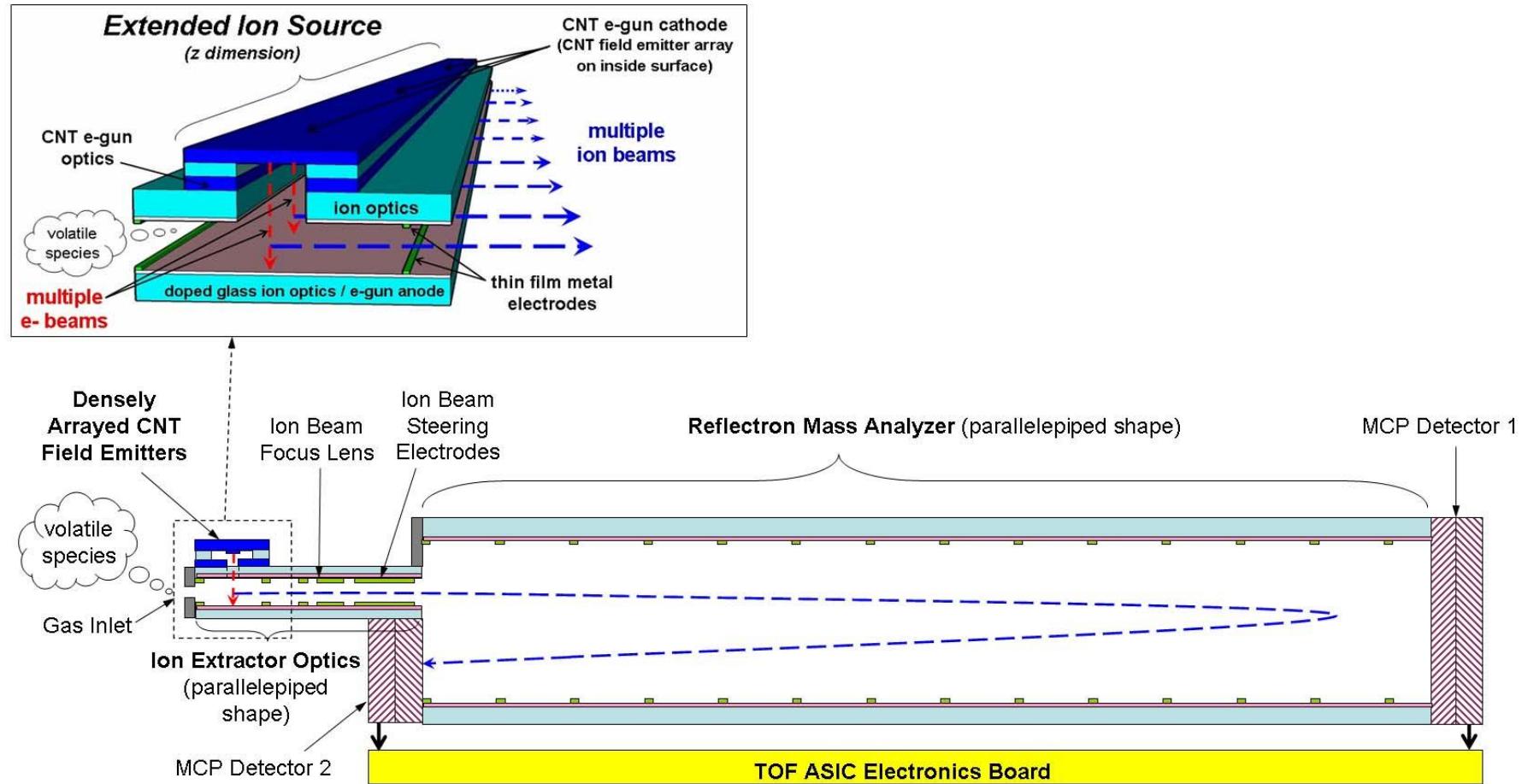


Objective	Instrument Target Measurements	Temp. Range (°C)
Origin of Volatiles	C, H, O, N, S-Volatiles	Atmospheric volatiles
		H <sub>2</sub> O, H <sub>2</sub> , CO <sub>2</sub> , CO, N <sub>2</sub> , SO <sub>2</sub> from regolith
		D/H ratio in H <sub>2</sub> O
		<sup>13</sup> C/ <sup>12</sup> C ratio of CO <sub>2</sub>
		<sup>15</sup> N/ <sup>14</sup> N ratio in N <sub>2</sub>
	Noble Gases	He, Ne, Ar, Kr, Xe, and Rn
		Isotope ratios ( <sup>3</sup> He/ <sup>4</sup> He, <sup>36</sup> Ar/ <sup>40</sup> Ar, <sup>129</sup> Xe/ <sup>136</sup> Xe)
	Organics	<sup>13</sup> C/ <sup>12</sup> C ratio in CO <sub>2</sub> from organics combustion
		Volatile hydrocarbons: methane, ethane, benzene, amines, alcohols, formaldehyde
Resources (ISRU)		Water-ice in regolith
		O <sub>2</sub>
		Reduced gasses such as HCN, NH <sub>3</sub> , and H <sub>2</sub> S
		<sup>3</sup> He relative abundance



# How the TOFMS works (old design)

***Electron impact ionization of volatile species followed by ion acceleration through reflectron.***

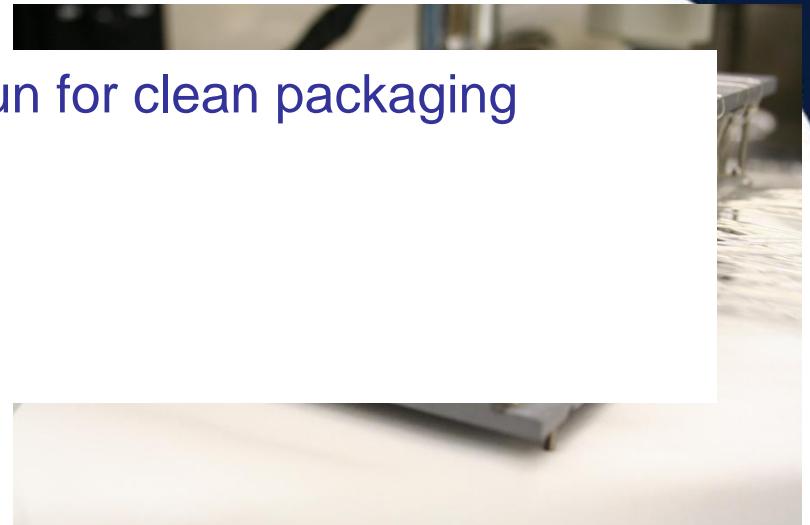


# Miniaturized Time-of-Flight Mass Spectrometer

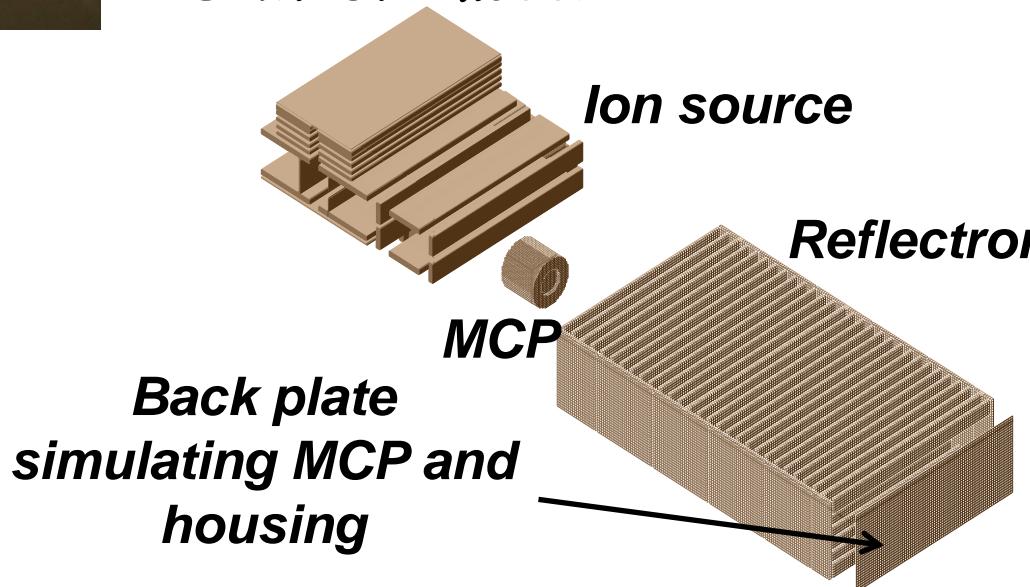


**MEMS Integration of CNT e-gun for clean packaging**

- Long lifetime
- Reduced current noise
- Reliable fabrication



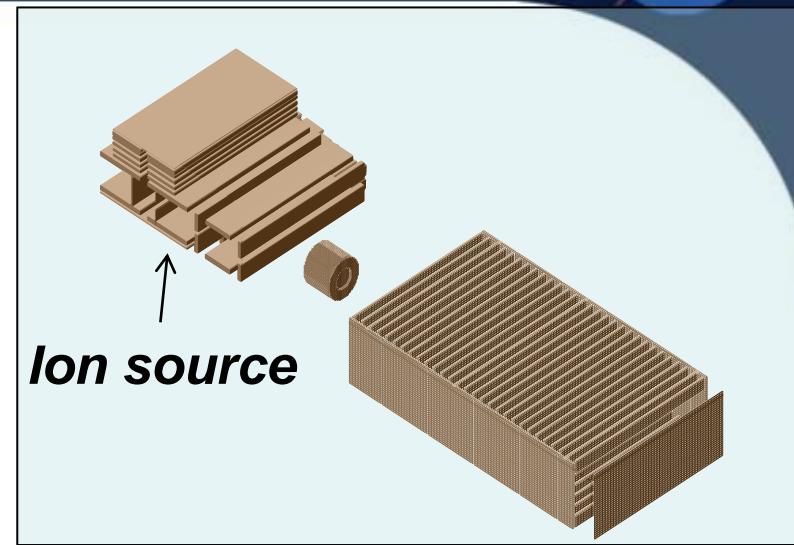
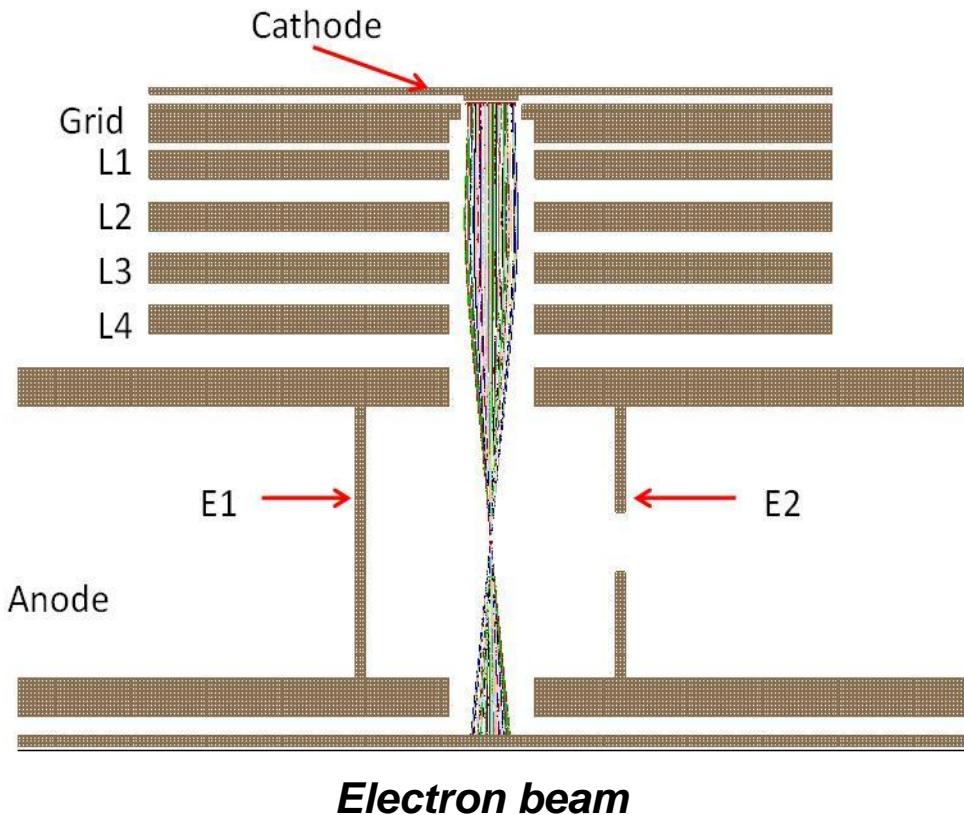
*SIMION model*



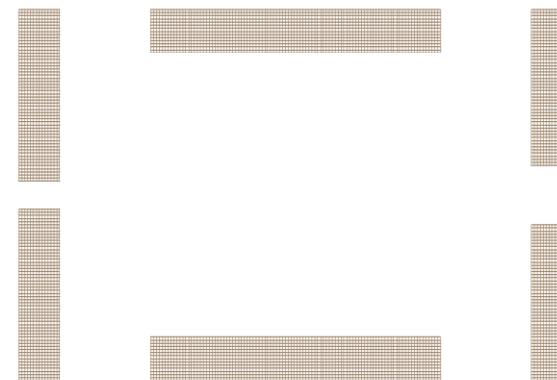
# NEW DESIGN with cross section



## ***Ion source cross-section***



F1                      Top                      F2

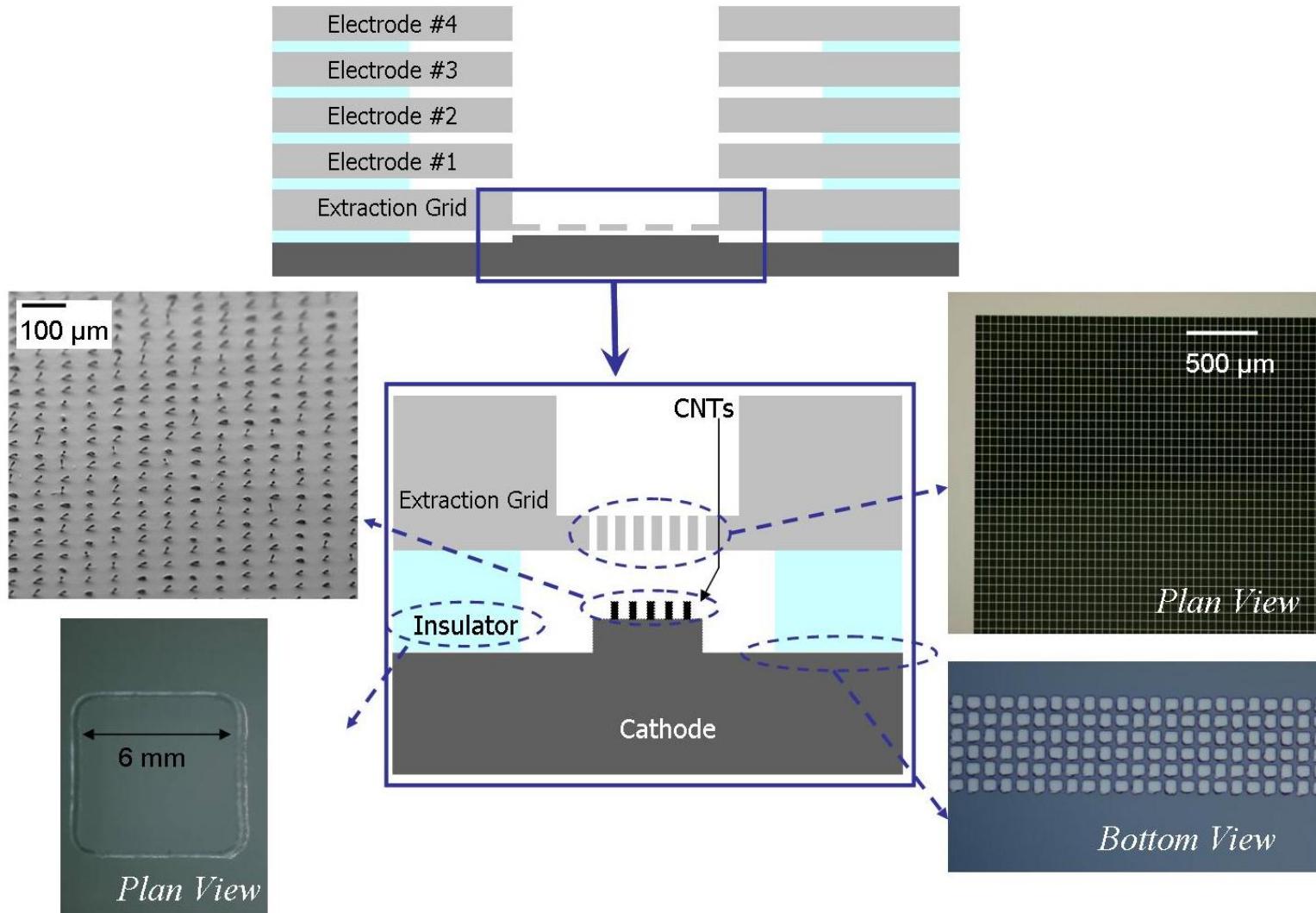


Bottom

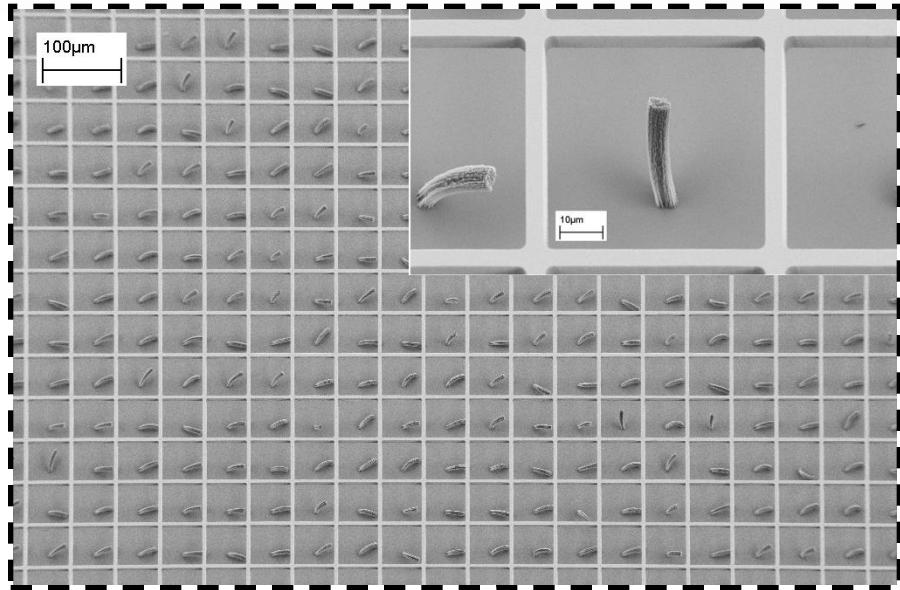
# Recent Work: Assembly and Integration



## Cathode-grid integration



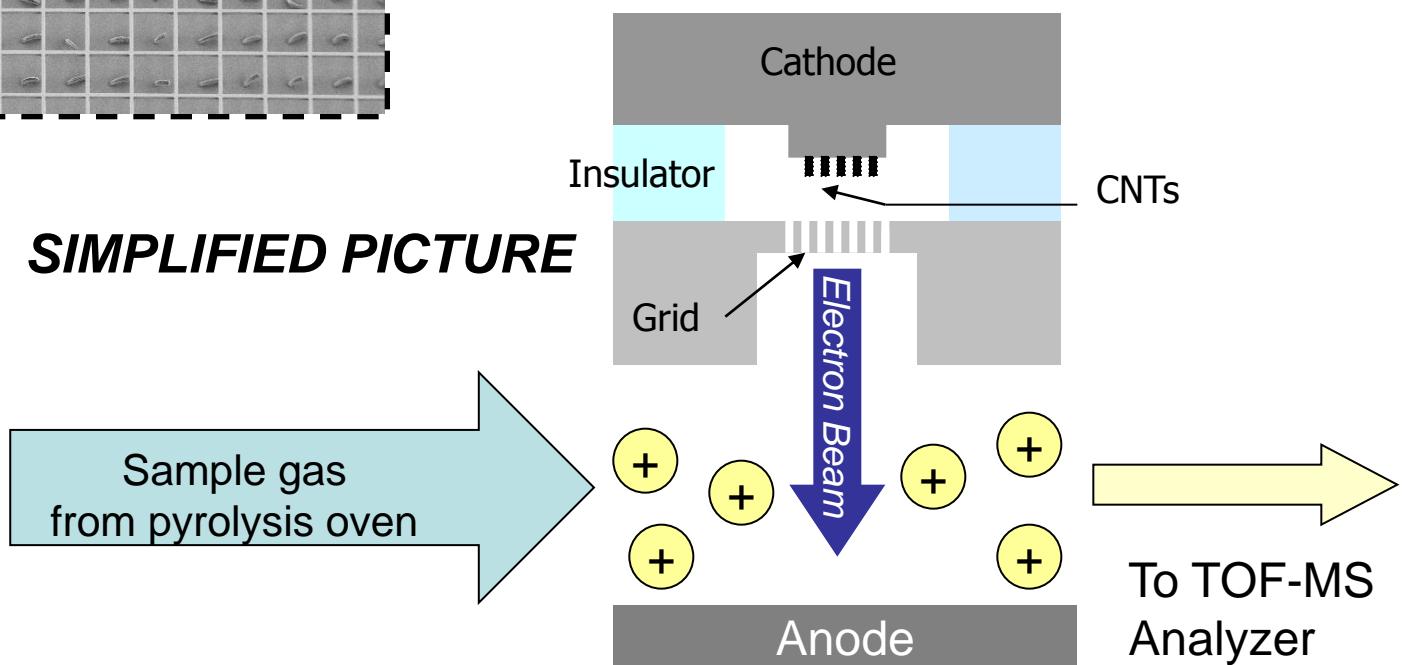
# Carbon Nanotube Electron Gun



## Carbon Nanotube field emitters

- Low power ( $100x <$  thermionic)
- Scalable for high sensitivity, redundancy
- Efficient ionization of the parent species
- Challenges: Lifetime and stability

## SIMPLIFIED PICTURE



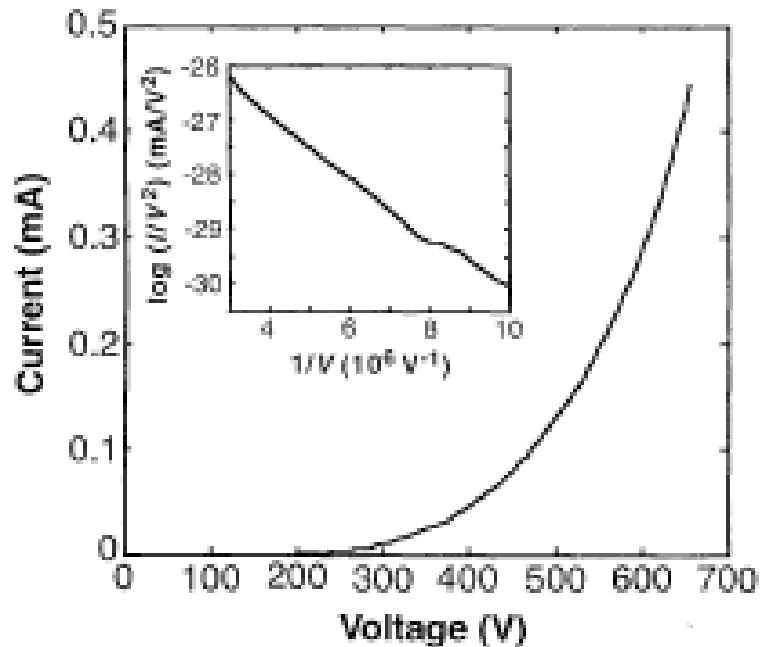
# Previous Work: Field Emission in CNTs



- Fowler-Nordheim Tunneling
  - $\beta$  =Field enhancement factor
  - Reported values of  $\beta$  vary from 400-1200 (for MWCNTs)

$$I = K_1 E^2 \exp\left(-\frac{K_2}{E}\right) \quad \beta = \frac{B\Phi^{3/2}}{K_2}$$

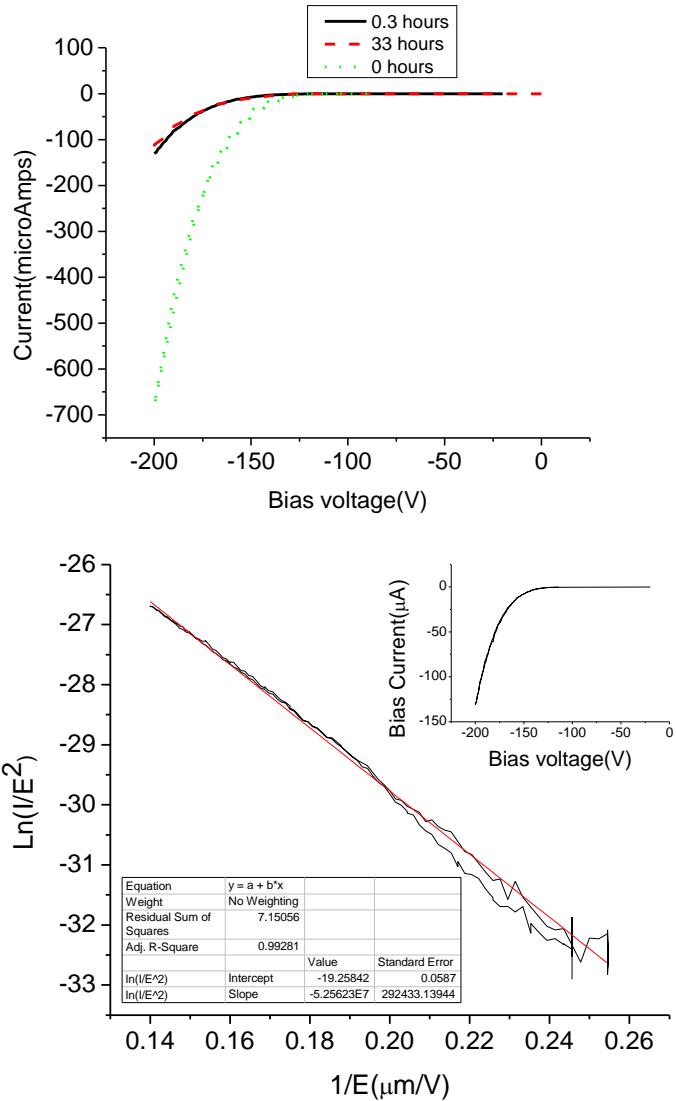
W. A. de Heer, A. Chatelain, D. Ugarte (1995)



# CNT E-gun for Mini Mass Spec

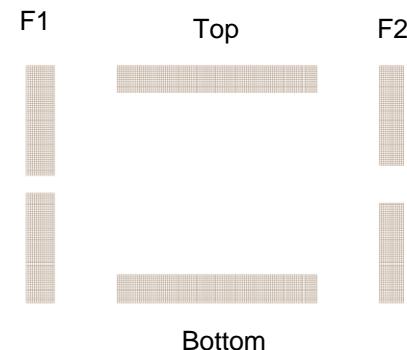
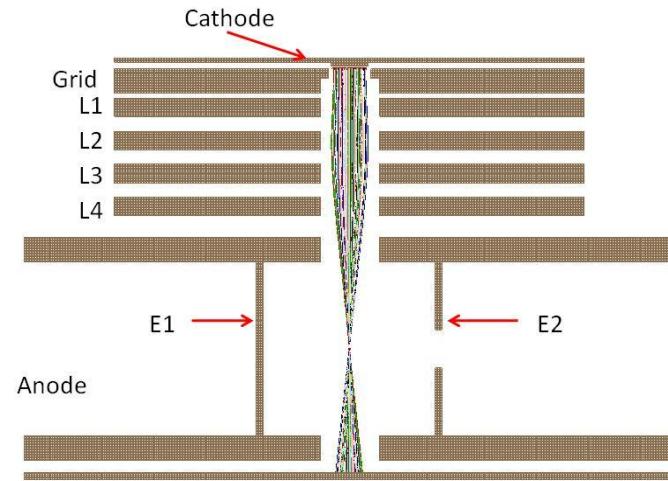


- Fowler-Nordheim behavior confirmed
  - $\beta \sim 1500$
- 100s of microamps of emitted current
- Recent tests show current persists after initial drop off and other devices have persisted for several hundred hours in high vacuum<sup>1</sup>



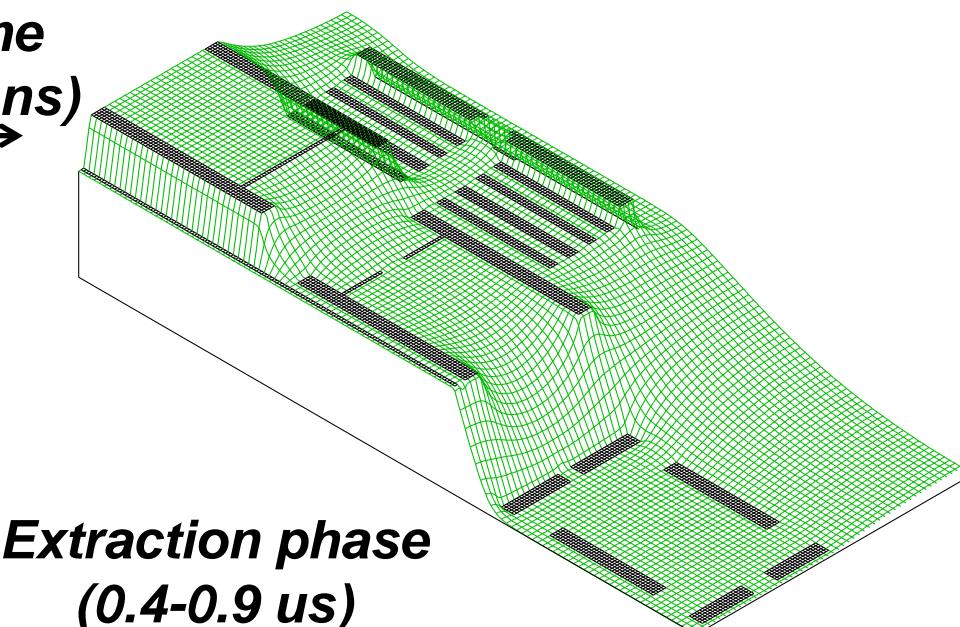
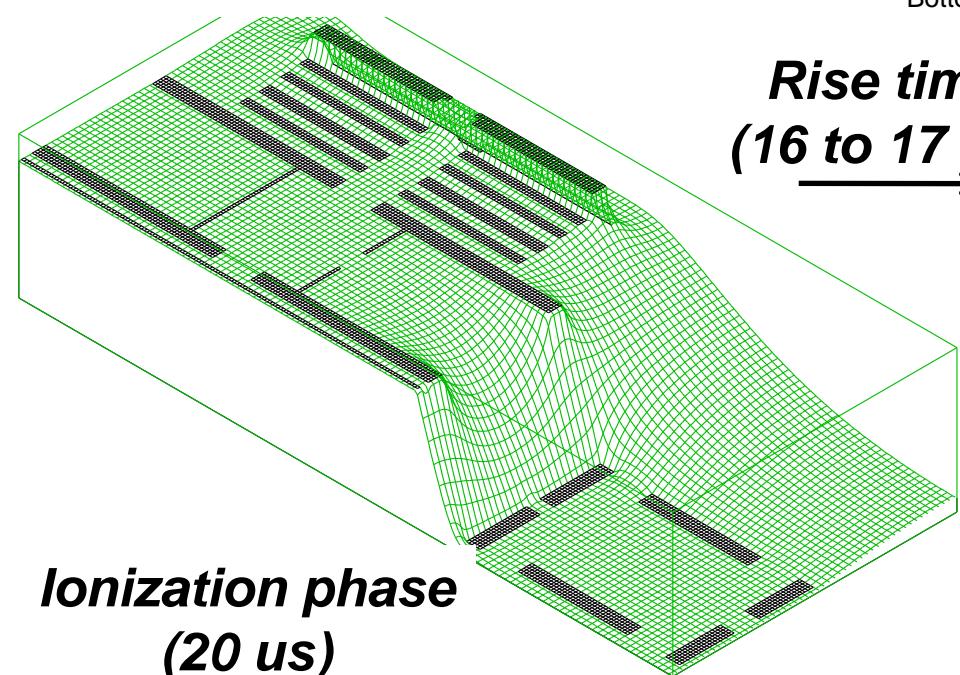
[1] S. A. Getty et al., "Effect of nitrogen gas on the lifetime of carbon nanotube field emitters for electron-impact ionization mass spectrometry," 2008, vol. 6959, pp. 695907-695907-10.

# SIMION Ion source simulation



Electrode	Ionization	Extraction
Anode	650	650
Cathode	580	580
Grid	780	780
L1	735	735
L2	670	670
L3	660	660
L4	650	650
E1	650	950
E2	650	650
F1=F2=Top=Bottom	0	0

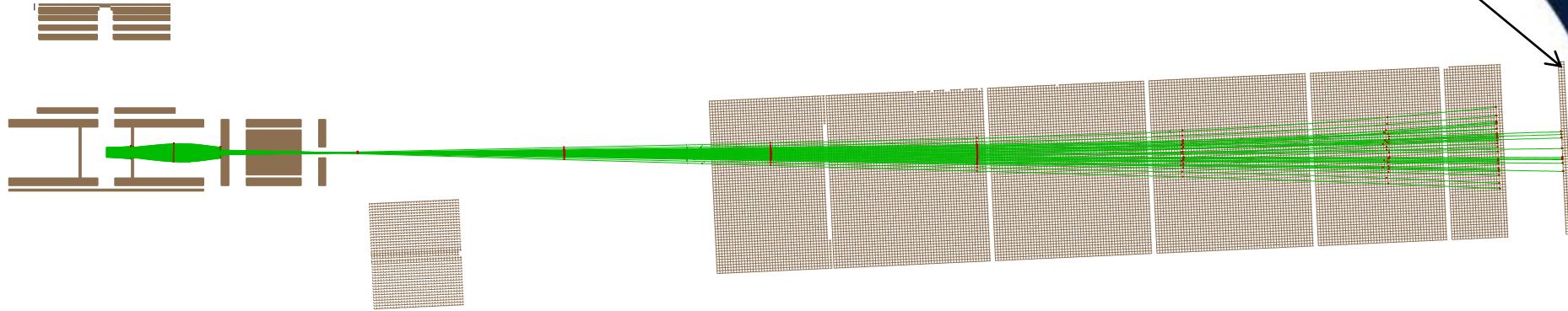
**Rise time**  
**(16 to 17 ns)**



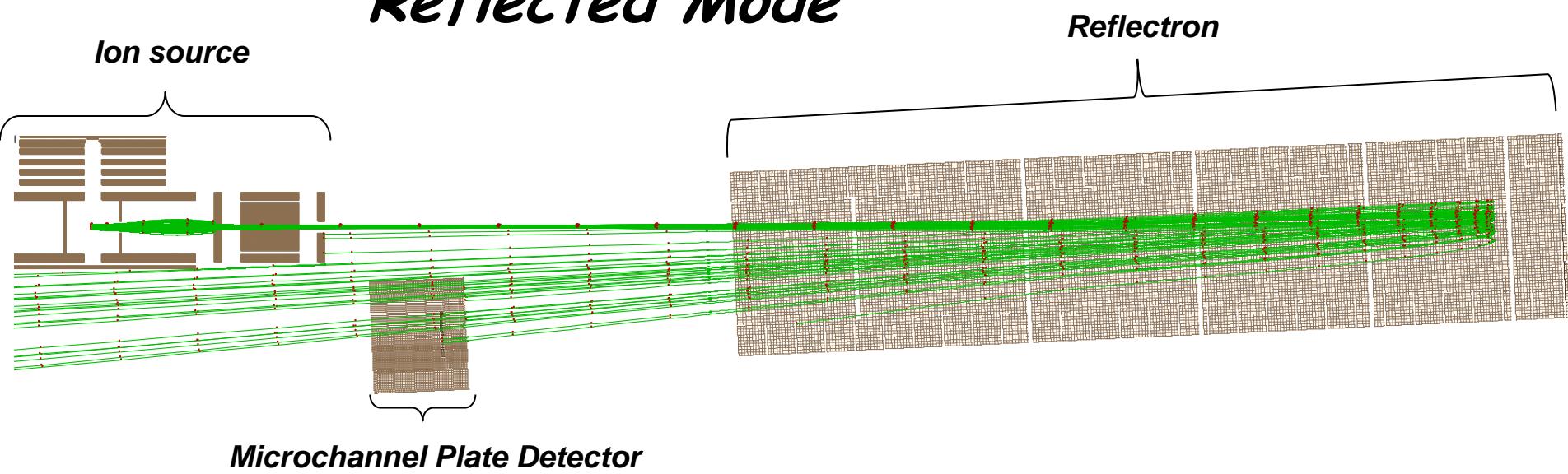
# Relectron operation modes



*Linear Mode*



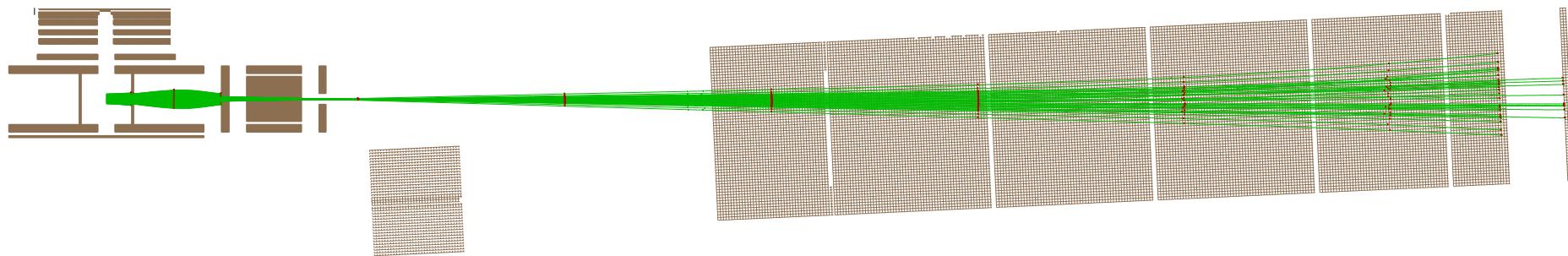
*Reflected Mode*



# SIMION vs. experiment



Simulated vs. experimental TOF



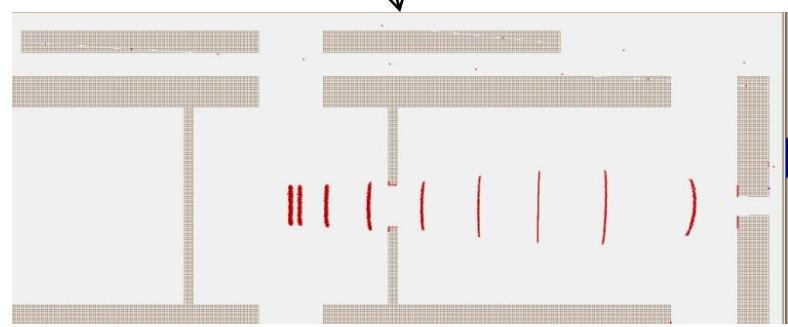
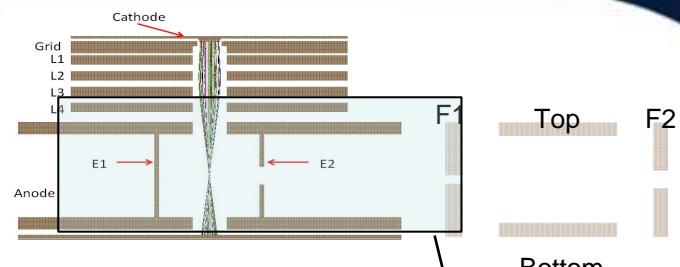
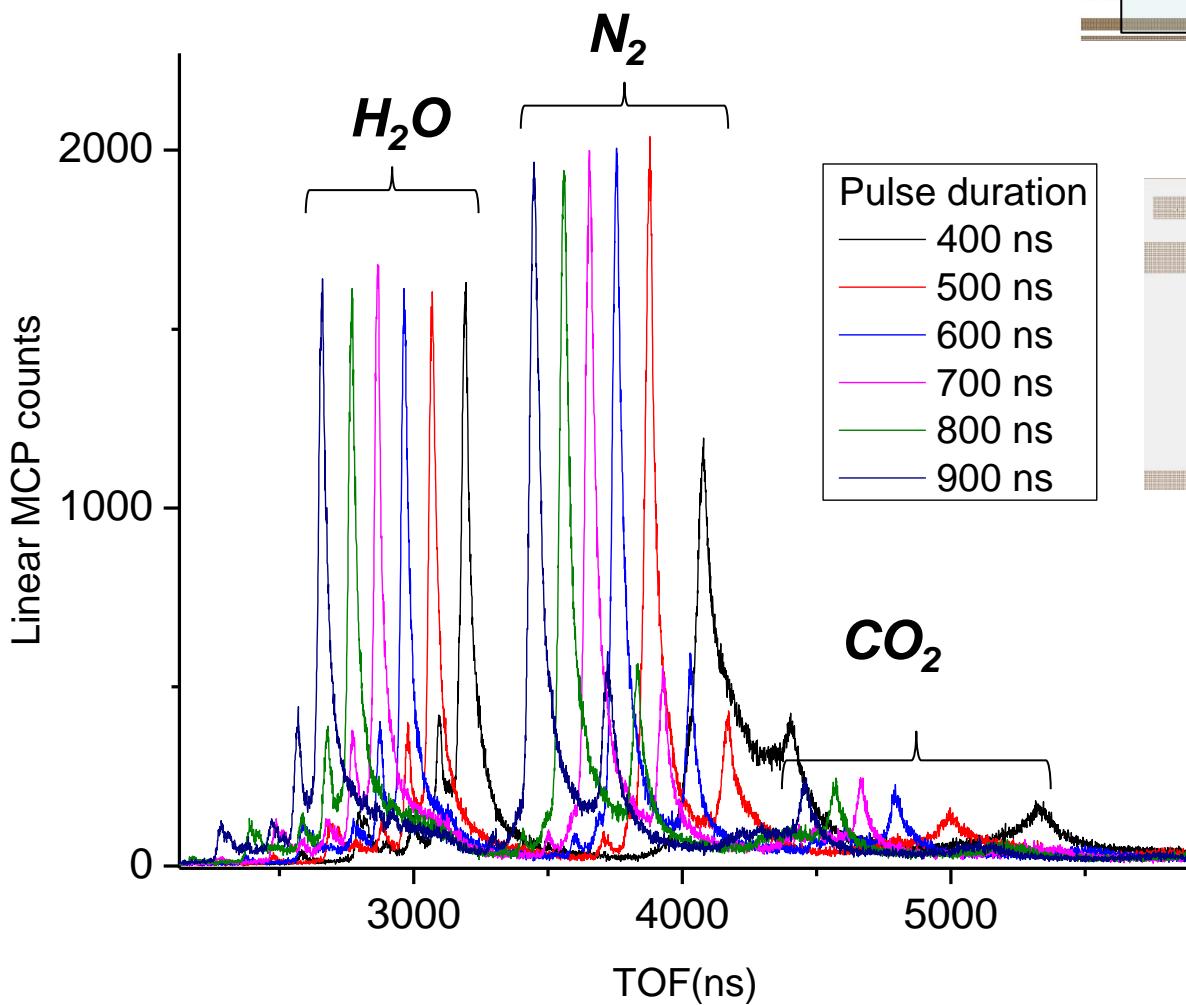
	H2O peak ( $\mu$ s)	N2 peak ( $\mu$ s)	CO2 peak ( $\mu$ s)
SIMION	3.15	3.91	4.91
Experiment	3.168	3.944	4.96

***SIMION model and experiment deviate by <50ns***

# Optimization of pulse duration

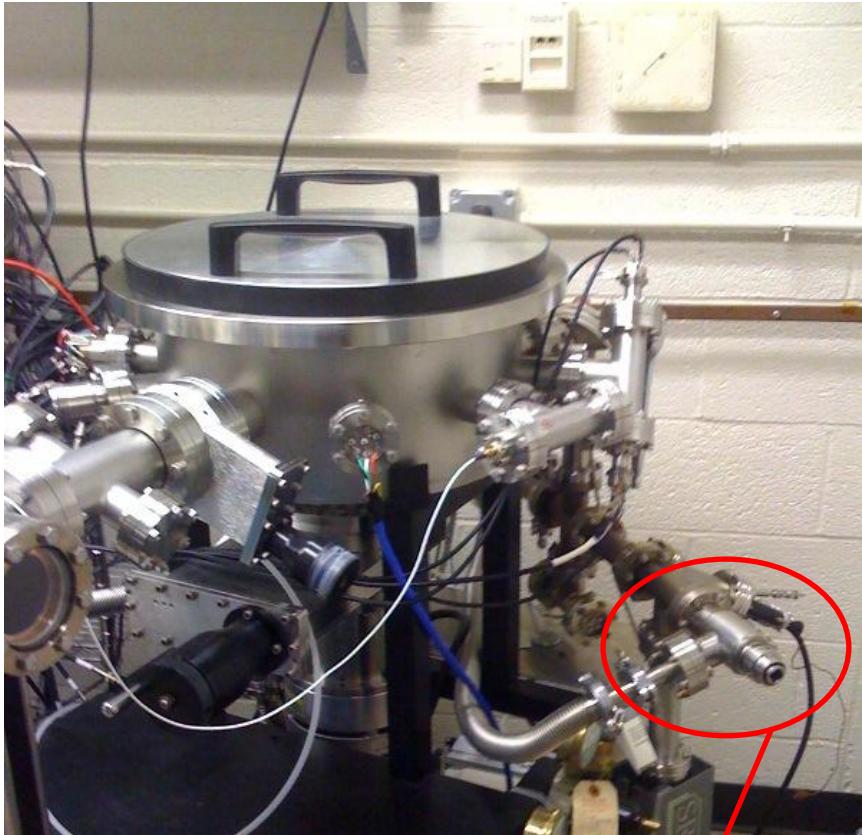


**300 V pulse**



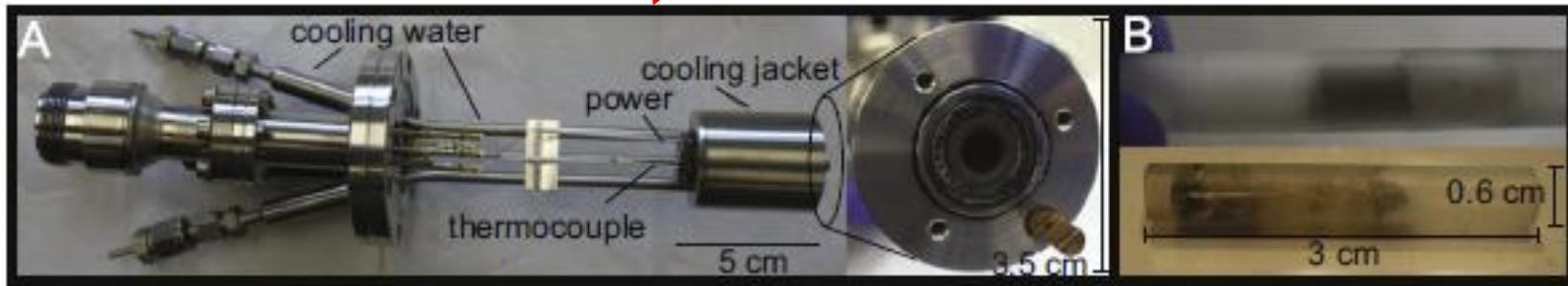
*Red time markers indicate  $CO_2$  ion position every 100 ns starting at t=0*

# Knudsen cell

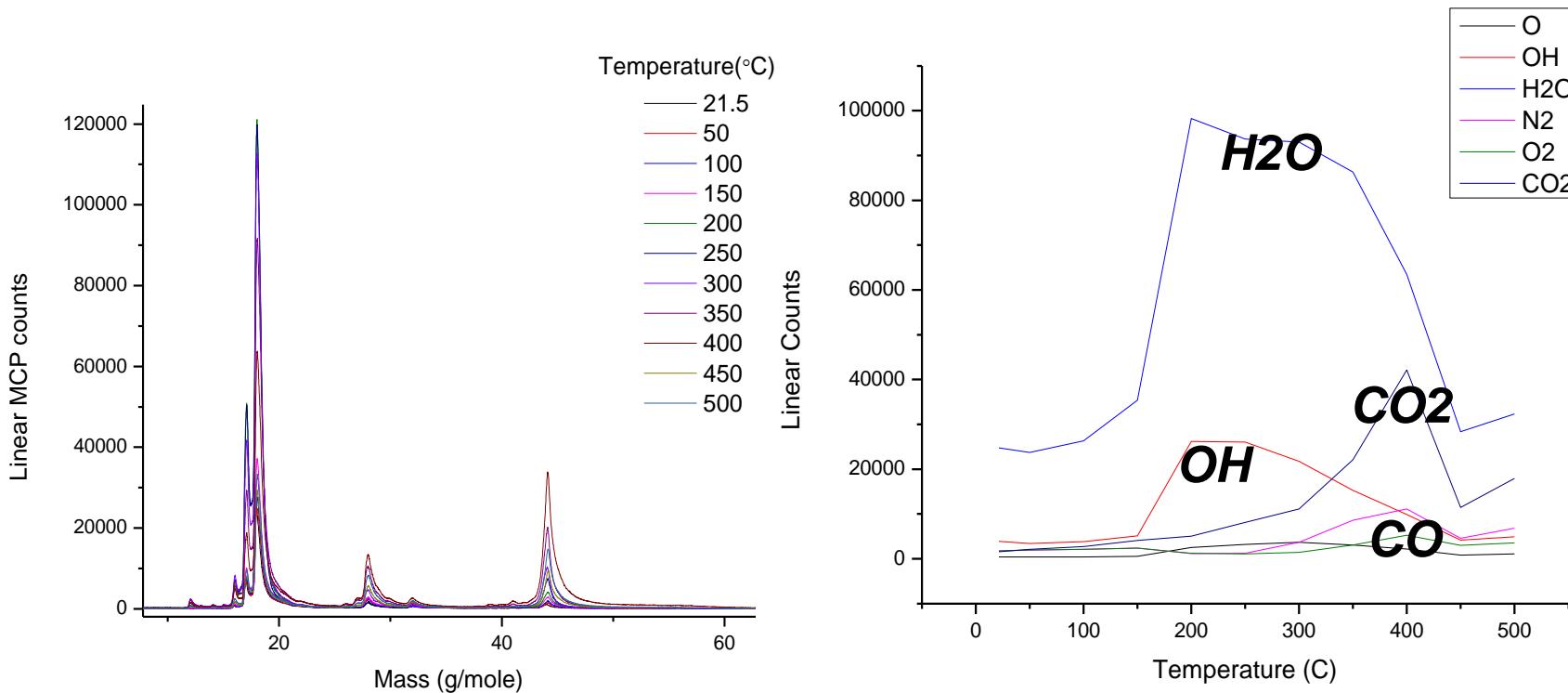


## *Knudsen cell testing:*

- Used for preliminary testing
- Heated up to 500 °C
- Sample packed into quartz tube

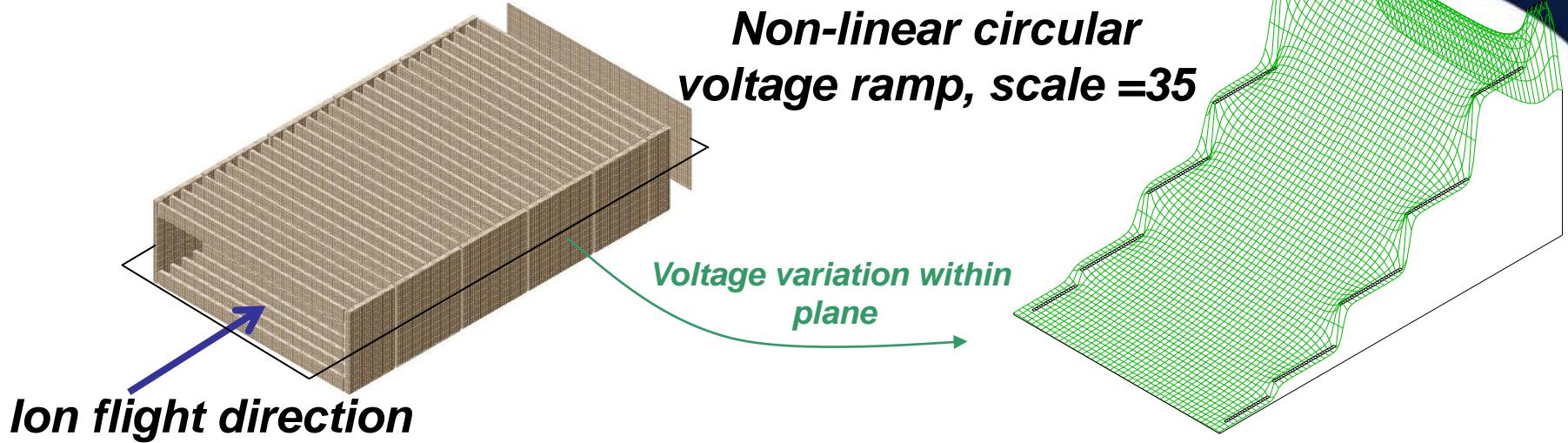


# MARS simulant heated to 500 °C using a Knudsen cell

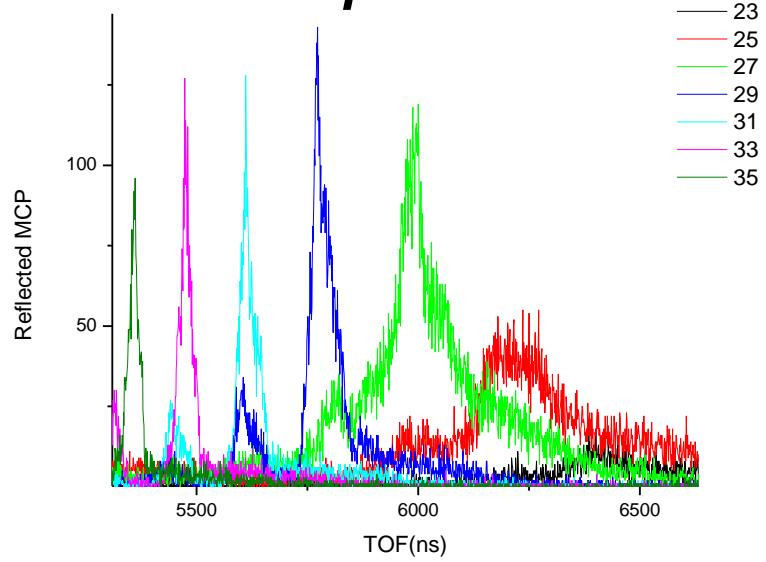
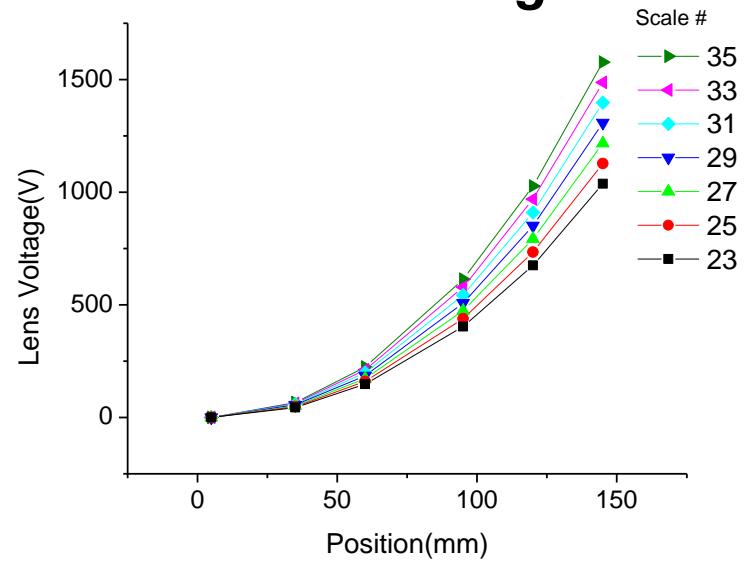


- **Mass accuracy to within 0.2 Daltons via calibration to  $H_2O$  and  $CO_2$**
- **$OH$ ,  $H_2O$  evolved at 200 °C while  $N_2/CO$ ,  $O_2$ ,  $CO_2$  evolved at 400 °C**

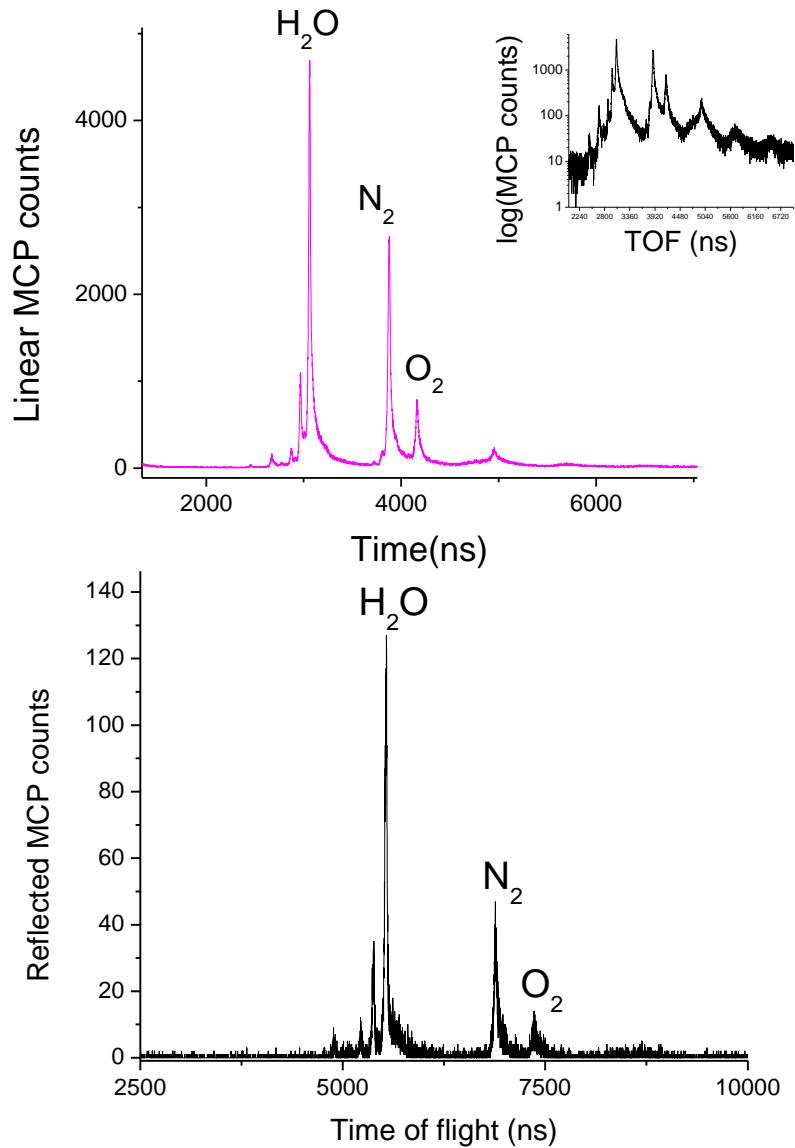
# Reflected mode spectra



**Higher scales -> Steeper voltage ramp ->  
highest mass resolution water peak**



# Comparison between linear and reflected mode



**Mass resolution ( $H_2O$ )= 130**  
**Sensitivity 3 counts per pulse**  
**7% transmission (SIMION)**

**Mass resolution ( $H_2O$ )= 130**  
**Sensitivity 0.1 counts per pulse**  
**0.3% transmission (SIMION)**

# Opportunities for TOF performance improvement:



Change	PROS	CONS
<ul style="list-style-type: none"><li>•Optimize emitter/grid geometry (increased current)</li><li>•Vertical pillar growth (increased current)</li><li>•Wider MCP detector under ion source</li></ul>	Increase sensitivity	
<ul style="list-style-type: none"><li>•Increase apertures in ion source</li></ul>		Decrease resolution
<ul style="list-style-type: none"><li>•Modify geometry of ion source</li></ul>		
<ul style="list-style-type: none"><li>•Time lag focusing through pulsed e-gun</li><li>•Improve field definition in reflectron w/ plates and fewer floating electrodes</li></ul>	Increase mass resolution	Decrease sensitivity
Use of other carbon based emitters	Improve emitter lifetime	
<ul style="list-style-type: none"><li>•Static → Dynamic</li><li>•Refine initial kinetic energy distribution</li></ul>	Better simulation	
Replace Knudsen cell with VAPoR oven	Integration with VAPoR	



# Acknowledgements

## GSFC Planetary Environments Lab

- **Daniel Glavin**
- Paul Mahaffy
- Charles Malespin
- Marvin Noreiga
- Vince Holmes
- Inge Ten Kate
- Dan Carrigan
- William Brinckerhoff
- Ray Bendt

## Instrument Electronics Development Branch

**Steve Feng**

## Honeybee Robotics

Eric Mumm

## GSFC Materials Engineering Branch

- **Stephanie Getty**
- **Greg Hidrobo**

## GSFC Detector Systems Branch

- **Carl Kotecki**
- **Nicholas Costen**
- **Larry Hess**
- Mary Li
- Patrick Roman
- Audrey Ewin

## Instrument Systems Branch

**Todd King**

*This work was supported by the ASTID program, NASA Innovation Fund, and the GSFC Internal Research and Development Program*